



**Beihang University**

## **Beihang University Micro-satellite System Research**

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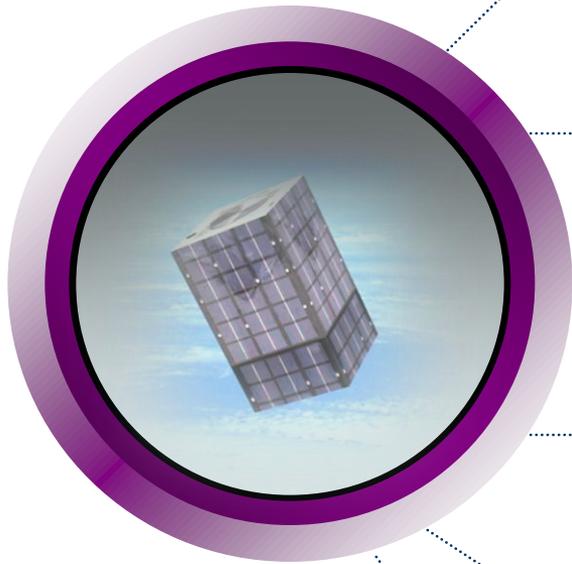
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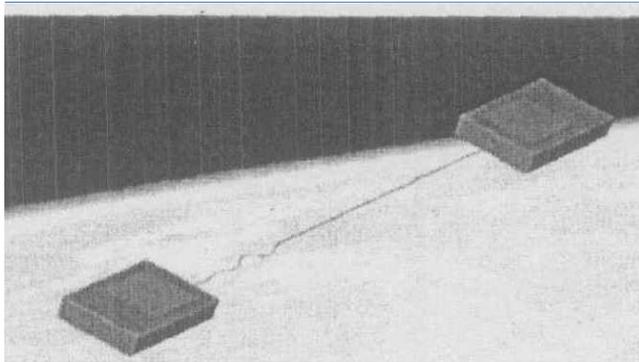
# 1. Background

## 1.1 University Micro-satellite Overview

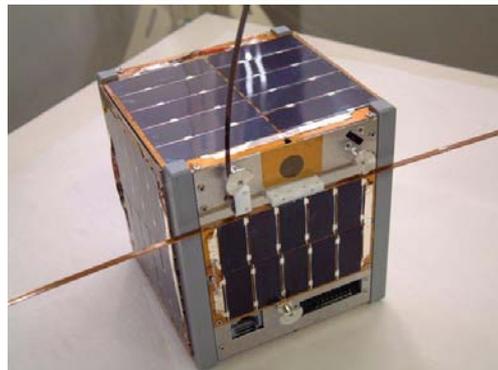
- ❖ University micro-satellites are becoming one of the key research field in future space exploration.
- ❖ Low cost spacecraft applied space exploration
- ❖ Educational Function. Influence faculty/student thinking and engineering skill activities



SSETI



PicoSat  
(Stanford University)



XI-V  
(Tokyo University)



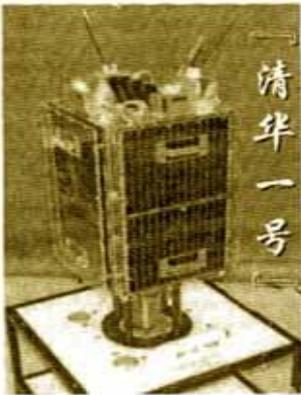
Delfi-C3  
(Delfi University of Technology)



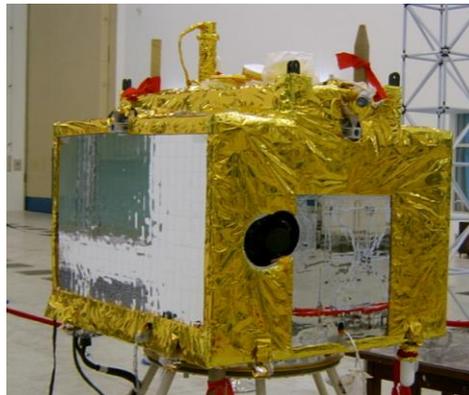
# 1. Background

## 1.2 University Micro-satellite Key Technologies:

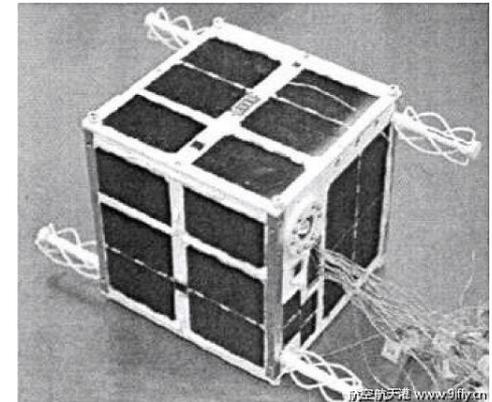
- New concept and new technology
- Advanced MEMS & system integration technology
- New energy sources research
- Micro-satellite constellations



**Tsinghua-1**  
(Tsinghua University)



**TS-1**  
(Harbin Institute of Technology)



**ZDPS-1**  
(Zhejiang University)



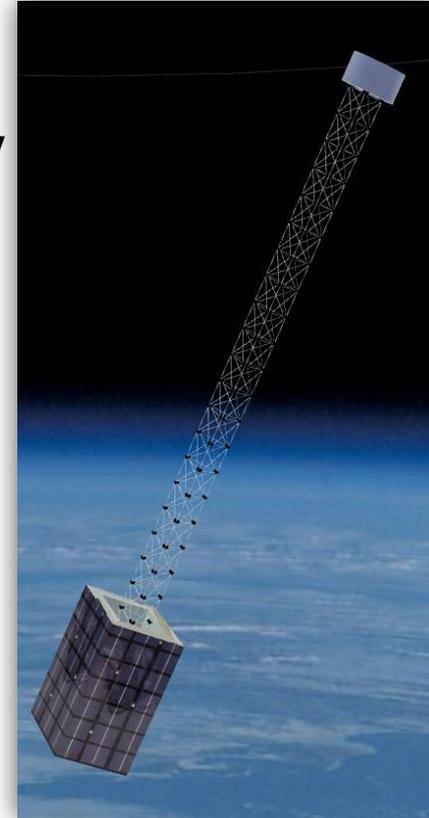
# 1. Background

## 1.3 BUAA-SAT Mission

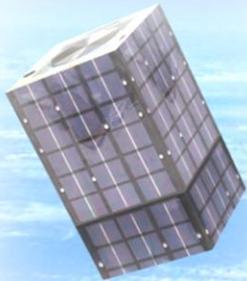
- ① Advanced micro-satellite system integration technology
- ① Coilable mast deployment technology
- ① Advanced integrated electrical system
- ① On-orbit imaging and on-board data compression
- ① Educational Function (students skill practice )

### BUAA-SAT supported by

- Beihang University Student Research Training Program
- Chinese Graduate Student Education Foundation



BUAA-SAT



## 2. Mission Overview

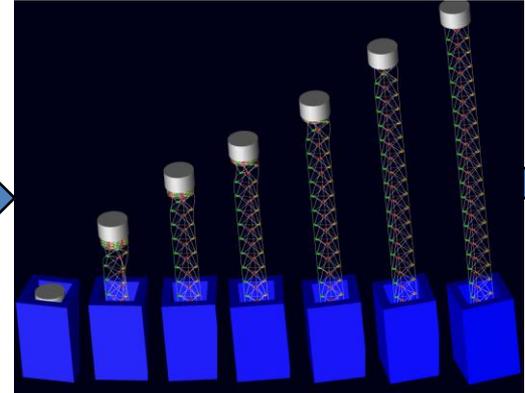
BUAA-SAT Mission Follows:



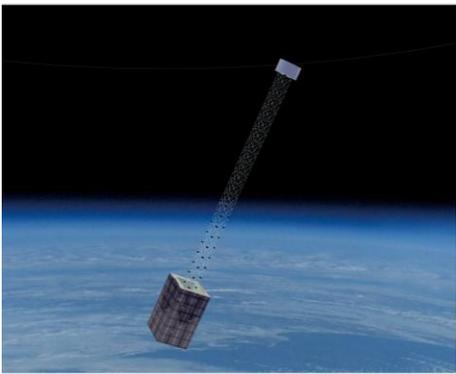
Launch



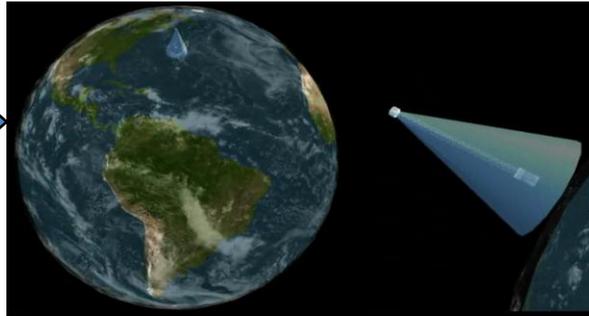
System Initialization  
& Damping



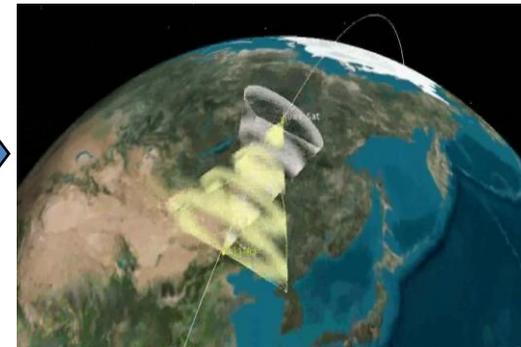
Coilable Mast Deploy



Gravity Gradient Stabilization



Taking Photo of Satellite  
& Observing the Earth



Communicating with  
Ground Station



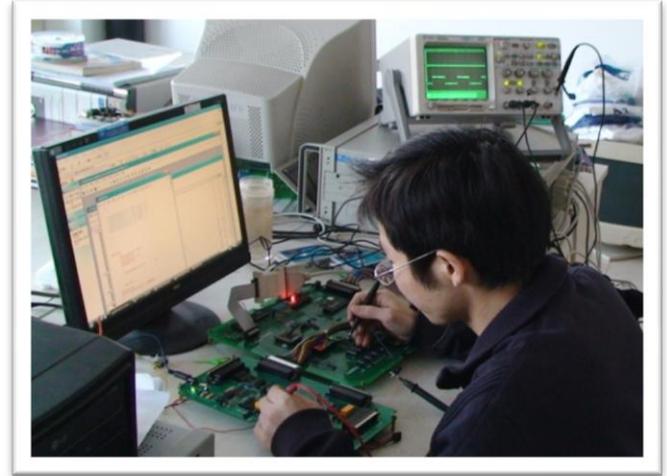
# 3. Spacecraft Overview

## 3.1 BUAA-SAT Design Philosophy

- **Modular Design**
- **Plug and Play (PnP)**
- **Low Cost: Commercial of the Shelf (COTS)**
- **Rapid-response application**

### System Features:

- ✓ **Standard hardware and software interface**
- ✓ **Allow change up until the last minute**
- ✓ **Easy to assemble and disassemble**
- ✓ **Easy to increase or decrease devices**
- ✓ **Reduce the cost and manufacture cycle**



# 3. Spacecraft Overview

## 3.2 BUAA-SAT Consists

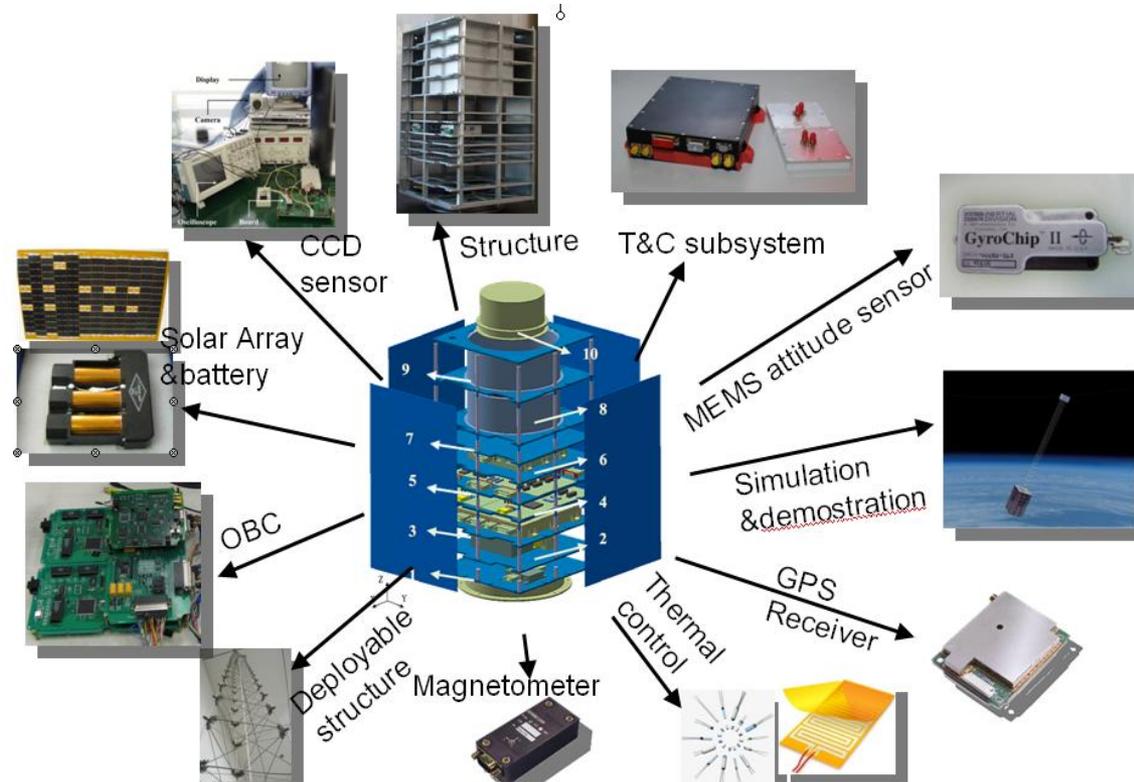
Subsystem	Characteristics
Structure	Layer-Rod Structure
Mechnism	2-meter long Coilable Mast
Thermal	Passive Control/MLI
Command and Data Handle	Base on ARM and CAN Bus
Electrical Power Subsystem	Body Mounted Solar Array, Lithium-ion batteries, Control Device
TT&C	S-band Transceiver UHF/VHF receiver
AOCS	Magnetometer, Coarse Sun sensor, GPS, Magnetrorque
P/L	Three Cameras and FPGA Control Board



# 3. Spacecraft Overview

## 3.3 System Characteristics

- **Dimension:** 300\*300\*500mm<sup>3</sup>
- **Weight:** 30~50kg
- **Orbit:** 600km Sun-synchronous
- **Power Supply:** 18W
- **Payload:** 3 Camera
- **Lifetime:** 3 months



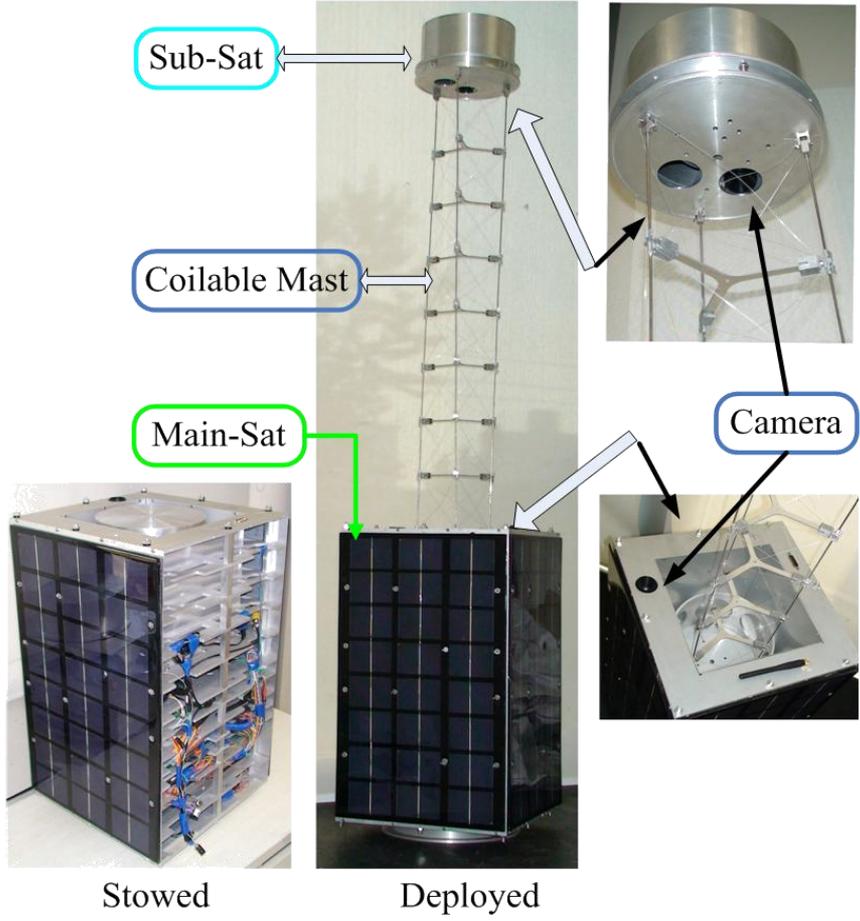
# 4. Key Technologies & System Tests

## 4.1 Structure

- Made of Duralumin alloy
- Layer-Rod Configuration

Main-satellite:  $300 \times 300 \times 500\text{mm}$

Sub-satellite:  $\Phi 100 \times 100\text{mm}$

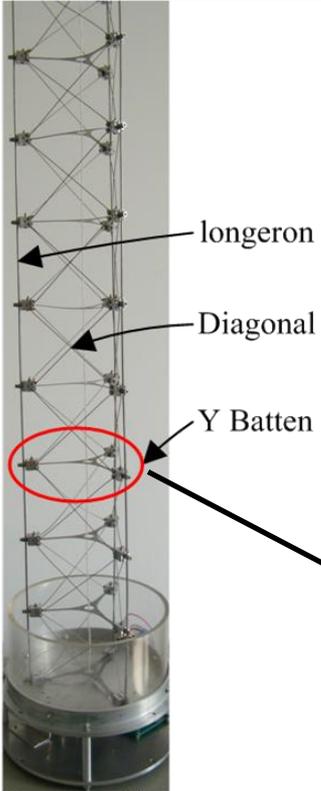


# 4. Key Technologies & System Tests

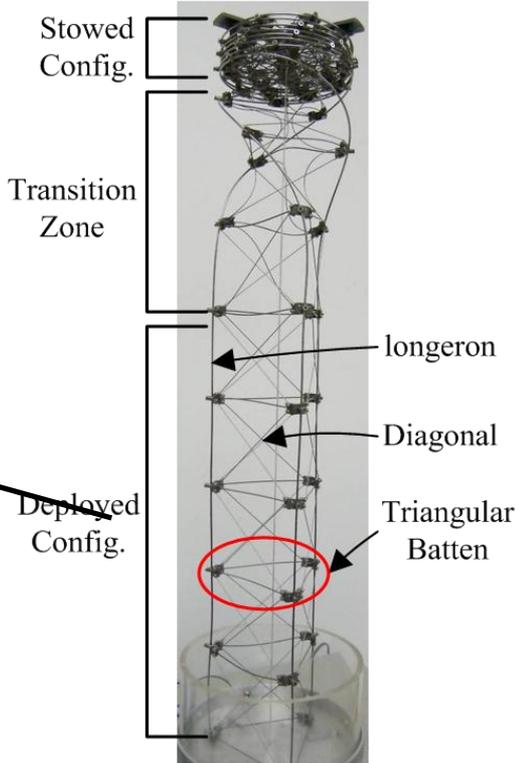
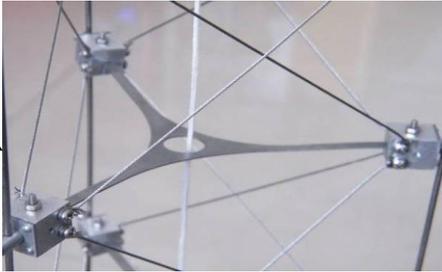
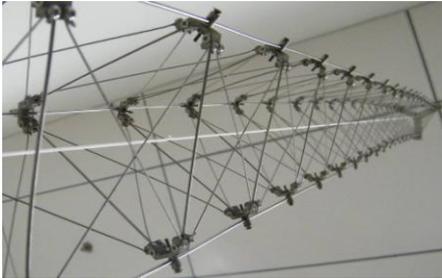
## 4.2 Coilable Mast

•Consists of longeron, diagonal and Batten.

Two Type According to the Batten style:  
1. triangular Batten  
2. "Y" type Batten



Y Batten  
Coilable Mast

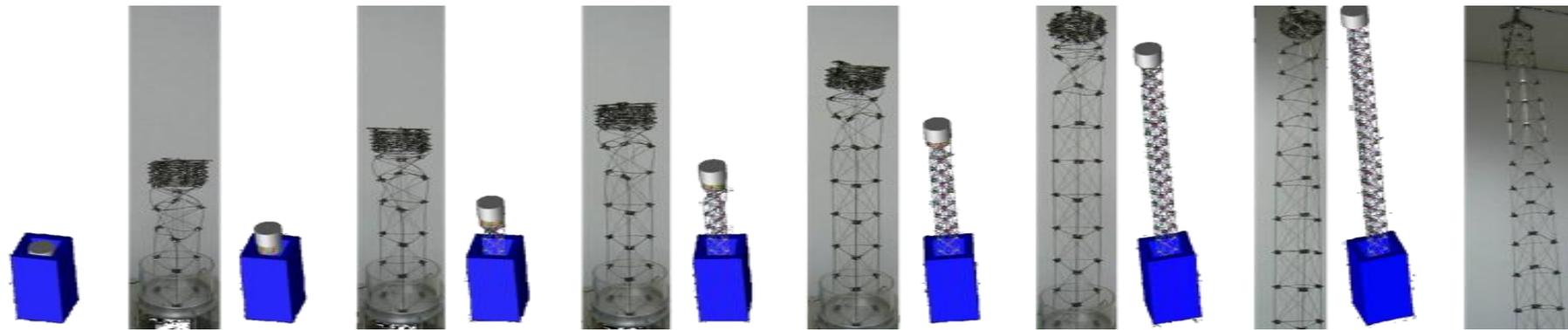
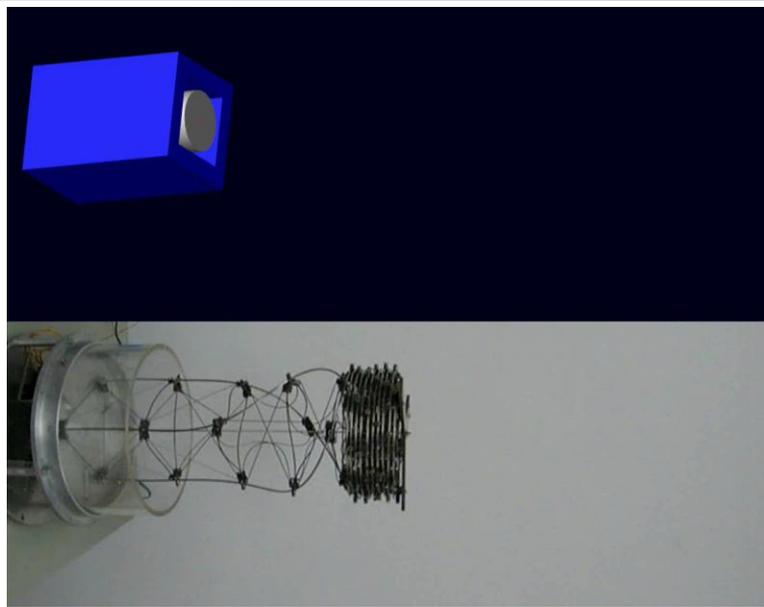


Triangular  
Coilable Mast



# 4. Key Technologies & System Tests

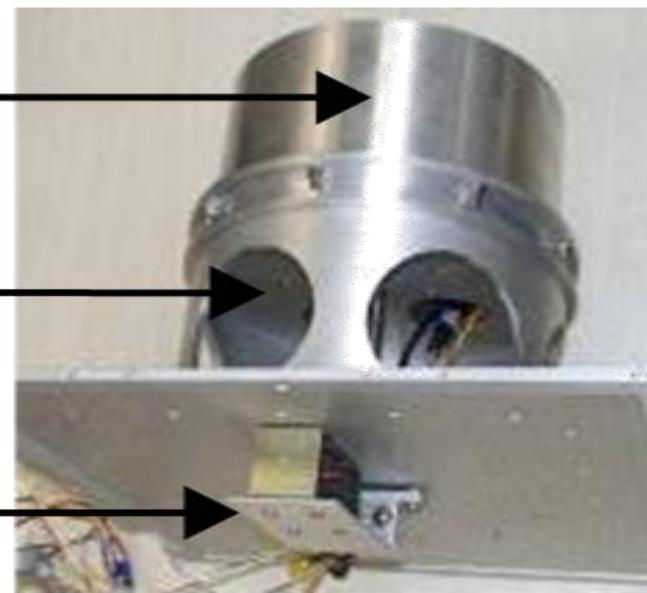
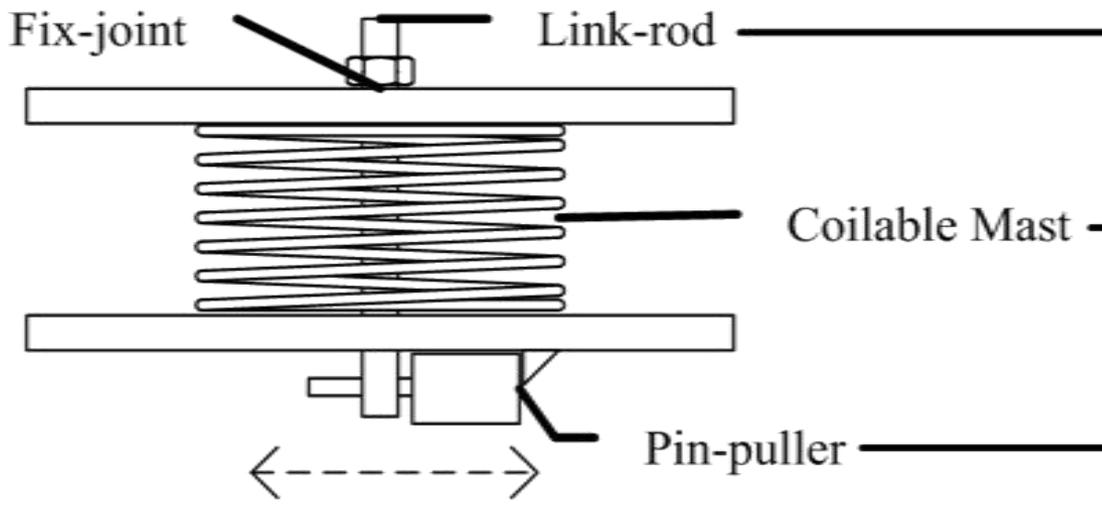
## Coilable Mast Verification by the test



# 4. Key Technologies & System Tests

## 4.3 Locking & Releasing Device

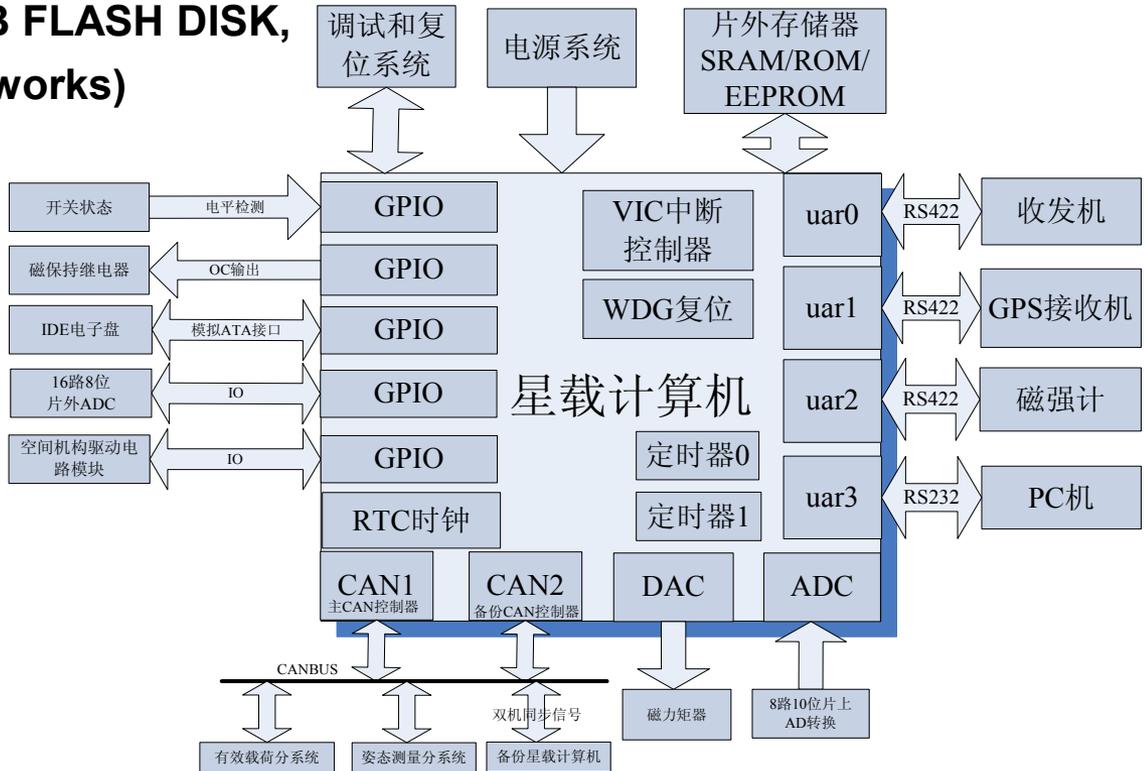
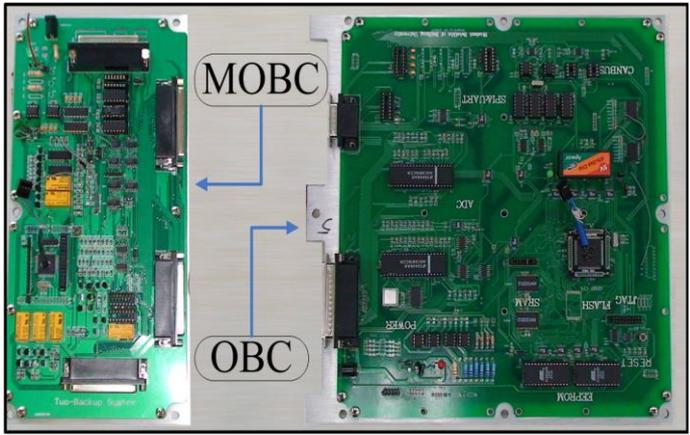
Lock and Release the mast



# 4. Key Technologies & System Tests

## 4.4 On-board computer

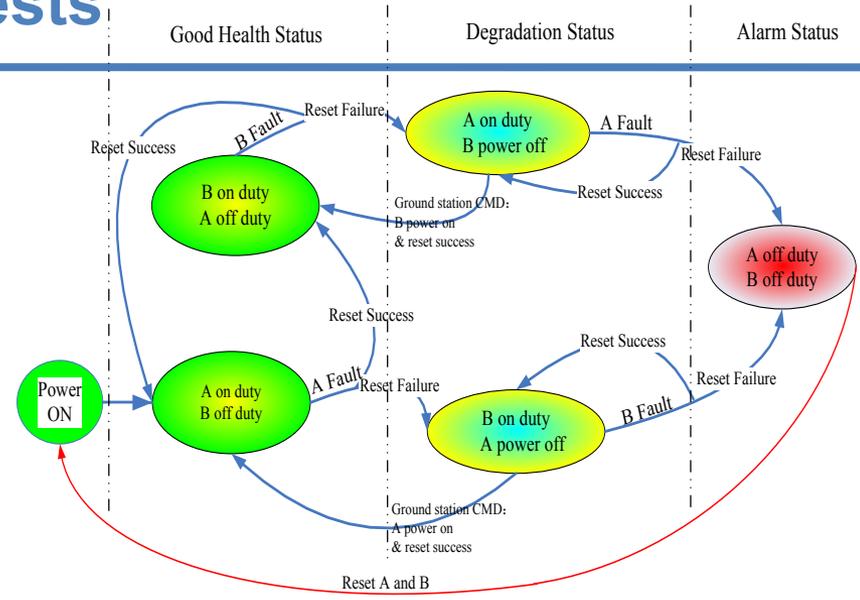
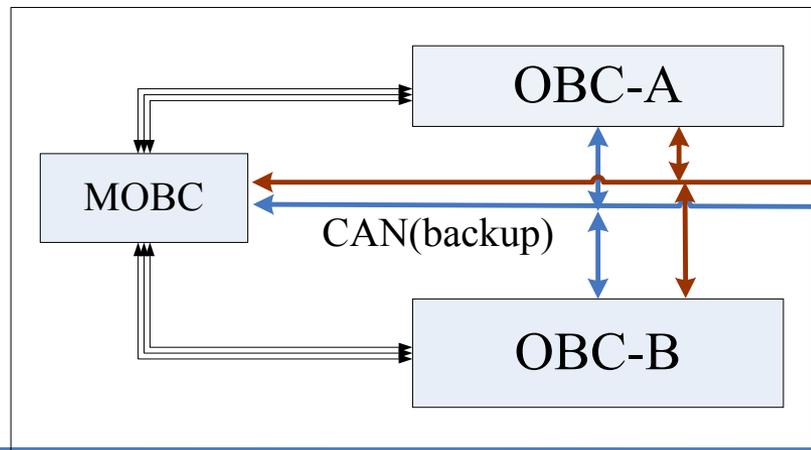
- Fault Tolerant Based on COTS
- ARM CPU @48MHz,
- 32KB PROM, 2MB SRAM, 128MB FLASH DISK,
- CAN bus 500kbps, I/F, RTOS(Vxworks)



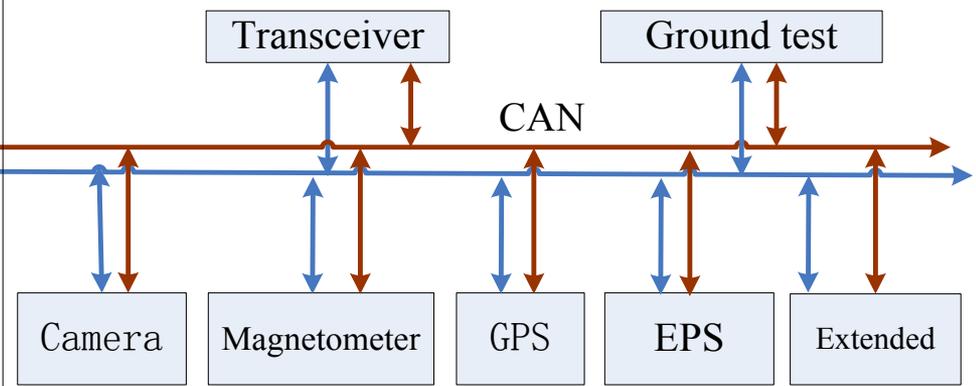
# 4. Key Technologies & System Tests

## On-board Data Handle (OBDH)

- OBDH subsystem consists of
- OBC-A
  - OBC-B (backup)
  - Monitor OBC (MOBC) monitor A and B.(2+1)



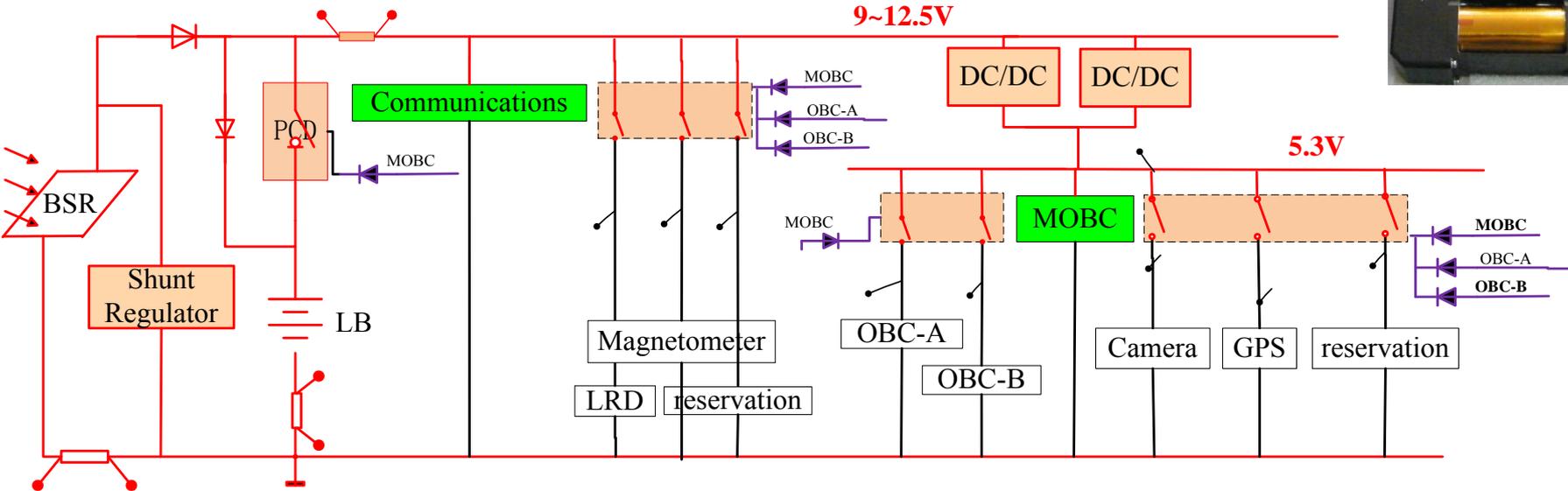
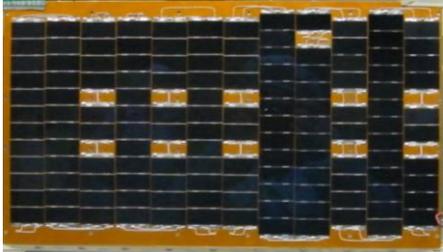
## On-Board Software FDIR Strategy



# 4. Key Technologies & System Tests

## 4.5 Electrical Power Subsystem (EPS)

- Body-mounted Solar Array
- Lithium-ion Batteries
- Power Control  
(unregulated voltage: 9~12.5V; regulated: 5.3V)

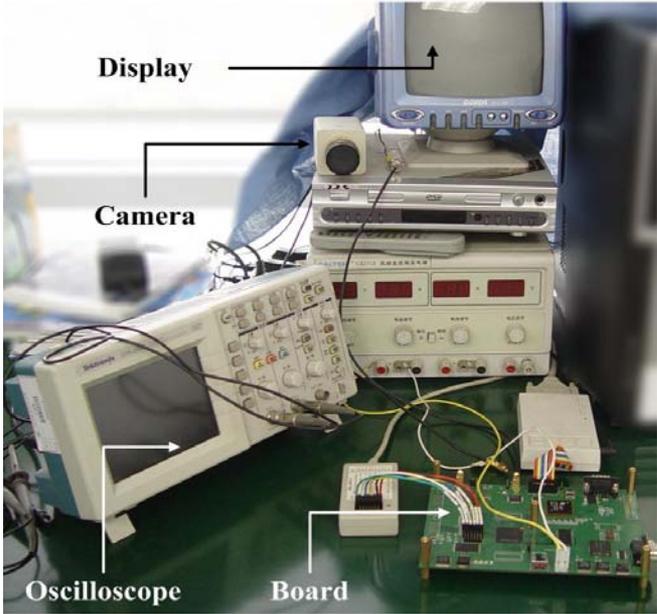
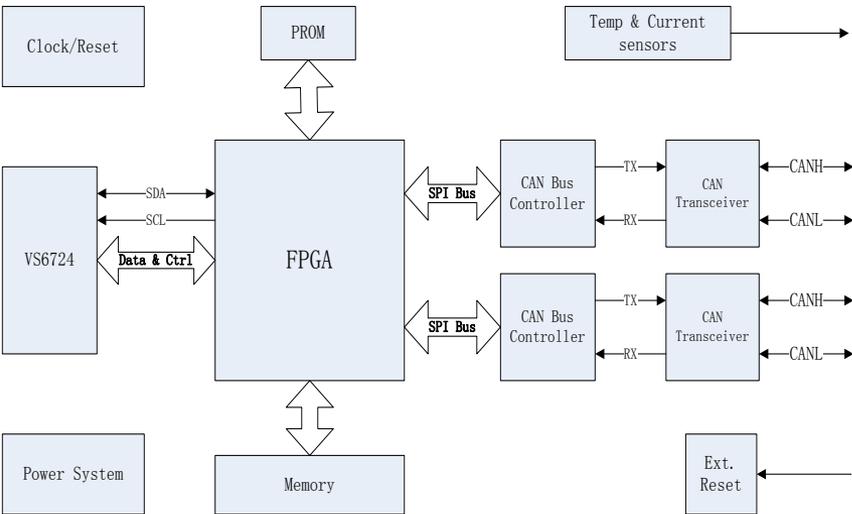


# 4. Key Technologies & System Tests

## 4.6 Payload Subsystem

P/L Consists of

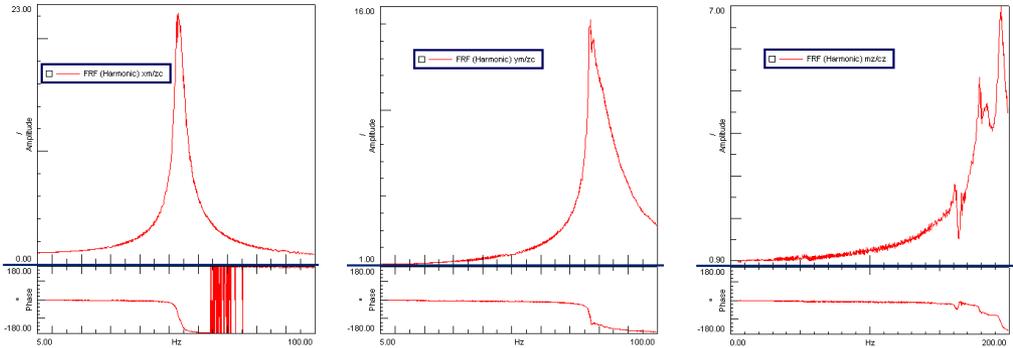
- 3 Cameras
- 1 Image Compression and Transmission Board  
(Image of Main-satellite, Sub-satellite, Mast and the Earth)



# 4. Key Technologies & System Tests

## 4.7 BUAA-SAT System Test(1)

### I. Structure Vibration Test



Vibration Testbed

### II. Random vibration test

	Require(Hz)	Test(Hz)	Analysis(Hz)
Transverse(x)	>12	53.5	46.058
Transverse(y)	>30	65.5	47.134
Longitudinal	>20	193.5	115.53



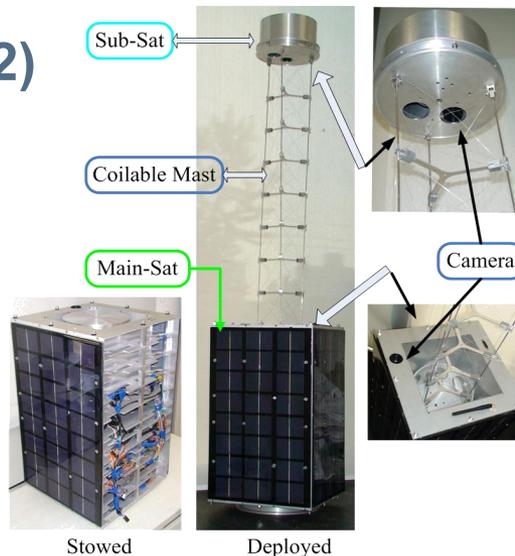
# 4. Key Technologies & System Tests

## 4.7 BUAA-SAT System Test(2)

### III. System Integrated Test

#### i). Function Test:

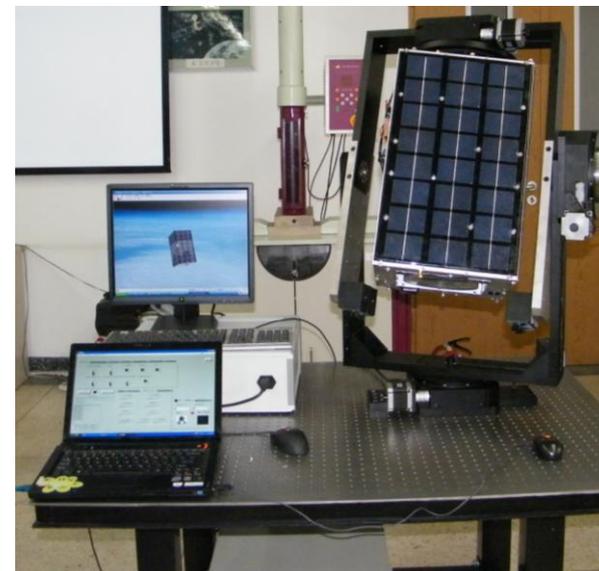
- Power Supply
- Transceiver
- P/L data Processing
- OBDH Management
- Attitude Determination
- Mast Deploying



System Integrated Test

#### ii). Platform Ground Test

- BUAA-SAT System
- 3-axis Rotate-Platform
- Visual Simulation System (VSS)
- Virtual Ground Station (VGS)



Platform Ground Test



# 4. Key Technologies & System Tests

## 4.7 BUAA-SAT System Test(3)

### IV. Vacuum Environment Test

#### i) Coilable Thermal Test

- **Operation Temperature:**  
-20°C~120 °C。
- **Vacuum Environment:**  
<10-7Pa Vacuum would be needed to test Cold Welding effect.



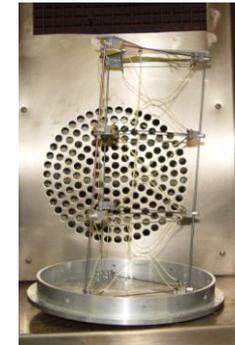
Heating Cage Facility



Vacuum Chamber



Thermal Vacuum Test

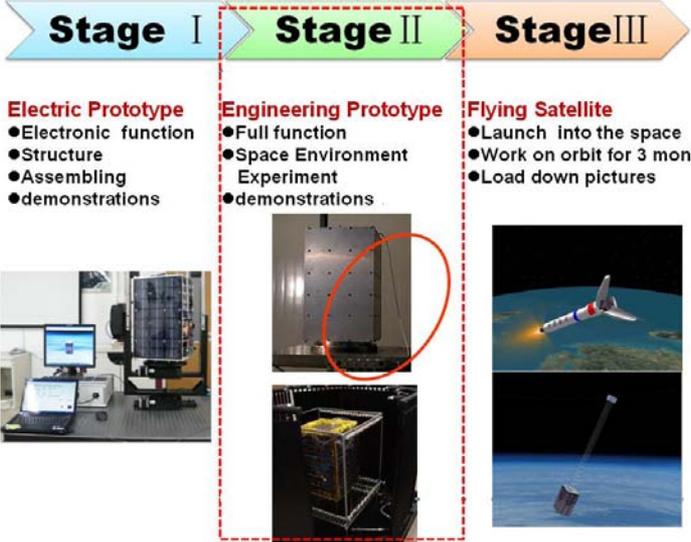


High/Low Temperature Test

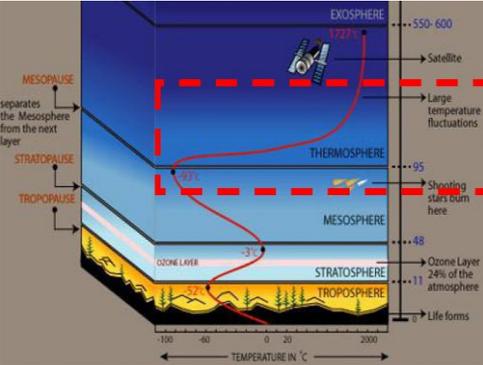


# 5. Future Work

- I. Work on Stage3:  
**BUAA-SAT Flight Model;  
 Aerospace Engineering Problems  
 Considered;**
- II. Looking for Launch Opportunities  
**Long March Piggyback  
 or Abroad Commerical Launch?**
- III. Finding Financial Assistance  
**???**
- IV. University Collaborations  
**BUAA-Picosat Space Exploration**



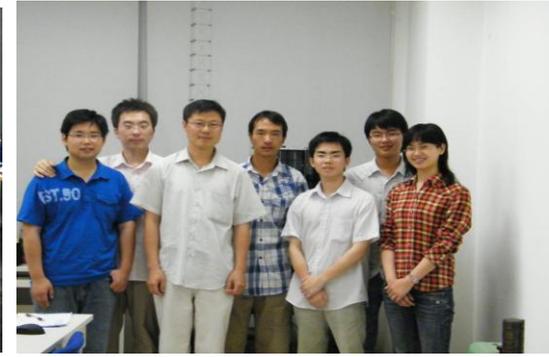
BUAA-Picosat



## 6. Conclusions

- **BUAA-SAT Ground Prototype Model Accomplished;**
- **Students' Aerospace Engineering Experience obtained;**
- **An Experienced R&D team has been trained;**

**Young Creative Talents  
Are Precious Wealth in 21 Century**





北京航空航天大学

BEIHANG UNIVERSITY

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**Thank you**

